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# Synchronizing Tracking Control for Flexible Joint Robots via Estimated State Feedback

In this paper, we propose a synchronization controller for flexible joint robots, which are interconnected in a master-slave scheme. The synchronization controller is based on feedback linearization and only requires measurements of the master and slave link positions, since the velocities and accelerations are estimated by means of model-based nonlinear observers. It is shown, using Lyapunov function based stability analysis, that the proposed synchronization controller yields local uniformly ultimately boundedness of the closed loop errors. A tuning gain procedure is presented. The results are supported by simulations in a one degree of freedom master-slave system. [DOI: 10.1115/1.1636197]

# 1 Introduction

This paper addresses the problem of synchronization of robot manipulators with flexible joints. It is assumed that not all the joint state variables are measured, e.g., link and rotor velocities are unknown, which is a very common situation when joint flexibility is present. In robotic systems synchronization is of great importance as soon as two robots have to cooperate. This cooperative behavior gives manoeuvrability and dexterity that can not be achieved by an individual system, e.g., multi finger robothands, multi robot systems, walking robots.

Typically robot coordination and cooperation, see [1,2], and [3], form important illustrations of the same goal, where it is desired that two or more mechanical systems, either identical or different, are asked to work synchronously. This is obviously a control problem that requires the design of suitable controllers to achieve the required synchronous motion. The problem of synchronization of robots, can be seen as tracking between the systems with some additional challenges that are not considered in tracking controllers.

Joint flexibility or joint clasticity, considerably affects the performance of robot manipulators [4], and is a major source of oscillatory behavior. To improve the performance of robot manipulators, joint flexibility has to be taken into account in the modelling and control of such systems. Joint flexibility can be caused by transmission elements such as harmonic drives, belts, or long shafts; and it can be modelled by considering the position and velocity of the motor rotor, and the position and velocity of the link. Therefore, the order of the dynamic model for a flexible joint is twice that of a rigid joint, consequently, the controllers are more complex than those for rigid joint robots. From a modelling point of view, two dynamic models for the flexible joint robot have been considered. In [5] an extended model for flexible joint robots is presented, which includes the full nonlinear dynamic interactions among joint flexibilities and inertial properties of links and actuators. If it is assumed that the kinetic energy of the electrical actuators is due only to their own rotor spinning, then a reduced model is obtained [6]. This reduced model satisfies the conditions of full state linearization and decoupling via static state feedback. Meanwhile it has been proved that the extended model is fully linearizable and decouplable via dynamic state feedback [7,8]. Other techniques like adaptive control, singular perturbations, and robust control have been investigated to design effective controllers for flexible joint robots (see [9,10]).

All the above mentioned controllers assume that all state variables are available, implying the presence of additional sensors in each joint, which in practice is difficult if not impossible. Besides the complexity in the implementation of measuring equipment, velocity measurements are often obtained by means of tachometers, which are often contaminated by noise, or moreover, velocity measuring equipment is frequently omitted due to the savings in cost, volume, and weight that can be obtained. To overcome this problem numerical differentiation, filters and the design of observers have been considered. Numerical differentiation and filters possess the advantage of simplicity in implementation. However they present a reduced bandwidth and in general there is no an analytical method to guarantee that the closed loop system will be stable. On the other hand, observers are in general modelbased, and thus require information about the system, and may thus be more difficult to implement. Nevertheless, model-based observers allow, in most cases, a stability proof and a methodology to tune the observer gains, which guarantee a stable closed loop system. In [11] a nonlinear observer based on pseudolinearization techniques has been proposed, a high gain observer is presented in [12], and a semiglobal nonlinear observer is designed in [13]. From an implementation point of view, it is desirable that controllers be designed to require few incasurements as possible. Examples of such philosophy are the controllers for flexible joint robots proposed in [14] and [15], which require only link and rotor position measurements.

In this work, the main goal is to ensure synchronization between two robots, where the robot for which the control will be designed (slave) has flexible joints, and the robot, whose trajectories are to be followed (master), may or may not have flexible joints. This goal is carried out assuming only link position measurements.

Consider two fully actuated robots with *n* joints each and working in a master-slave scheme, such that the subindexes *m*, *s* identify the master and slave robot. Assume that the master robot is driven by a torque  $\tau_m(\cdot)$ , that in the ideal case, ensures convergence of the link position and velocity  $q_m$ ,  $\dot{q}_m$  to a desired trajectory  $q_d$ ,  $\dot{q}_d$ . However, the torque  $\tau_m$ , the dynamic model and parameters of the master robot, as well as the link velocity and acceleration  $\dot{q}_m$ ,  $\ddot{q}_m$ , are not available for design of the slave controller  $\tau_s(\cdot)$ . Also we assume that the slave rotor position  $\theta_s$ , and the slave link and rotor velocities and accelerations  $\dot{q}_s$ ,  $\ddot{q}_s$ ,  $\ddot{\theta}_s$ ,  $\ddot{\theta}_s$  are not measured. Therefore, the slave controller  $\tau_s$ , that is to be designed such that the link variables  $q_s$ ,  $\dot{q}_s \in \mathbb{R}^n$  synchronize with the variables  $q_m$ ,  $\dot{q}_m$ , can only depend on link position measurements of both robots, i.e.,  $q_m$ ,  $q_s$ , and estimated or re-

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constructed values of  $\theta_s$ ,  $\dot{\theta}_s$ ,  $\dot{q}_m$ ,  $\ddot{q}_m$ ,  $\dot{q}_s$ ,  $\ddot{q}_s$ . Notice that the goal is to follow the link trajectories of the master robot  $q_m$ ,  $\dot{q}_m$ , and no the desired trajectories  $q_d$ ,  $\dot{q}_d$  for the master robot, which may not be realized due to model uncertainties or disturbances, e.g., noise, unknown loads.

A similar setup and goal have been considered in [16] for the case of rigid joint robots. In [16] semiglobal uniform, ultimate boundedness of the synchronization system is proved and a relationship between the controller gains and the ultimate error bounded is obtained. Throughout this paper for matrices the induced norm  $||A|| = \sqrt{\lambda_{max}(A^TA)}$  is considered, with  $\lambda_{max}(\cdot)$  the maximum eigenvalue. For any positive definite matrix A,  $A_m$  and  $A_M$  denote its minimum and maximum eigenvalue, respectively.

This paper is organized as follows, Section 2 presents a reduced model for the flexible joint robot and some of its properties. A nominal controller, assuming that all signals  $\theta_s$ ,  $\dot{\theta}_s$ ,  $q_m$ ,  $\dot{q}_m$ ,  $q_s$ ,  $\dot{q}_s$  are available, is introduced in Section 3. Section 4 presents a version of the nominal controller based on estimated values for  $\theta_s$ ,  $\dot{\theta}_s$ ,  $\dot{q}_m$ ,  $\dot{q}_s$ . The closed loop system formed by the slave robot and the synchronization controller is obtained in Section 5, and its stability analysis is presented in Section 6. Section 7 summarizes a tuning gain procedure for the gains in the controller and observers. A simulation study is presented in section 8. Some conclusions are given in Section 9.

#### 2 Reduced Dynamic Model of the Flexible Joint Robot

The dynamic model of a flexible joint robot can be obtained by extending the procedures already used for rigid robots [5]. Consider a flexible joint robot, with n rigid links, all joints being flexible, revolute, and actuated by electrical drives. Let  $q_i \in \mathbb{R}^n$  be the link positions and  $\theta_i \in \mathbb{R}^n$  be the rotor positions, as reflected through the gear ratios, the subindex i = m, s identifies the master (m) and slave (s) robot. The difference  $q_{ij} - \theta_{ij}$  is the j-th joint deformation, in view of small deformations, joint elasticity is modelled as a linear spring. The rotors of the motors are modelled as balanced uniform bodies having their center of mass on the rotation axis, so that the inertia matrix and the gravity term in the dynamic model are independent from the motor position  $\theta_i$ . Assuming that the motion of the rotors can be considered as pure rotations with respect to an inertial frame, then the kinetic energy of each rotor is due to its own spinning. Therefore, the inertial coupling between links and rotors can be neglected and a reduced dynamic model is obtained (see [6]). Following [6] and the Lagrangian approach [17], we obtain the dynamic model:

$$M_{i}(q_{i})\ddot{q}_{i} + N(q_{i},\dot{q}_{i}) + K_{i}(q_{i}-\theta_{i}) = 0 \quad i=m,s$$
(1)

$$J_i \ddot{\theta}_i + K_i (\theta_i - q_i) + B_{v,i} \dot{\theta}_i = \tau_i$$
<sup>(2)</sup>

$$N(q_{i}, \dot{q}_{i}) = C_{i}(q_{i}, \dot{q}_{i})\dot{q}_{i} + g_{i}(q_{i})$$
(3)

where the symmetric positive definite inertia matrix  $M_i(q_i) \in \mathbb{R}^{n \times n}$ , the Coriolis and centrifugal term  $C_i(q_i, \dot{q}_i) \dot{q}_i \in \mathbb{R}^n$ , and the gravity term  $g_i(q_i) \in \mathbb{R}^n$  are all related to the rigid links,  $J_i \in \mathbb{R}^{n \times n}$  is the constant diagonal inertia matrix of the motors,  $K_i \in \mathbb{R}^{n \times n}$  is the constant diagonal matrix of the joint stiffness,  $B_{v,i} \in \mathbb{R}^{n \times n}$  is the diagonal positive definite viscous friction coefficient matrix, and  $\tau_i(\cdot)$  is the *n*-vector of torques supplied by the motors. The following properties are satisfied by the model (1)–(3)

- If the matrix  $C_i(q_i, \dot{q}_i) \in \mathbb{R}^{n \times n}$  is defined using the Christoffel symbols [18], then the matrix  $\dot{M}_i(q_i) 2C_i(q_i, \dot{q}_i)$  is skew symmetric.
- In addition, for the previous choice of the matrix  $C_i(q_i, \dot{q}_i)$ , the Coriolis and centrifugal term  $C_i(q_i, \dot{q}_i)$  can be written as:

$$C_{i}(q_{i},\dot{q}_{i}) = \begin{bmatrix} \dot{q}_{i}C_{i1}(q_{i}) \\ \vdots \\ \dot{q}_{i}C_{in}(q_{i}) \end{bmatrix}$$
(4)

where  $C_{ij}(q_i) \in \mathbb{R}^{n \times n} j = 1, ..., n$  are symmetric matrices. It follows that

$$C_{i}(q_{i},x)y = C_{i}(q_{i},y)x$$

$$C_{i}(q_{i},z+\alpha x)y = C_{i}(q_{i},z)y + \alpha C_{i}(q_{i},x)y$$
(5)

for any scalar  $\alpha$  and for all  $q_i$ ,  $x, y, z \in \mathbb{R}^n$ .

• The matrices  $M_i(q_i)$ ,  $C_i(q_i, \dot{q}_i)$ ,  $B_{v,i} \in \mathbb{R}^{n \times n}$  are bounded

$$0 < M_{im} \leq ||M_i(q_i)|| \leq M_{iM} \quad \text{for all } q_i \in \mathbb{R}^n \tag{6}$$

$$\|C_i(q_i, x)\| \leq C_{iM} \|x\| \quad \text{for all } q_i, x \in \mathbb{R}^n$$
(7)

$$0 < B_{v,im} \leq ||B_{v,i}|| \leq B_{v,iM} \tag{8}$$

#### **3** Nominal Feedback Controller

Consider that (1)-(3) define the dynamics of two robots working under a master slave scheme. Based on inverse dynamics computation and using De Luca and Lucibello [8], the slave robot with model (1)-(3) can be fully linearized and decoupled via the static feedback control law:

$$\tau_{s} = K_{s}(\theta_{s} - q_{s}) + B_{v,s}\dot{\theta}_{s} + J_{s}K_{s}^{-1}[M_{s}(q_{s})v(t) + \alpha(q_{s}, \dot{q}_{s}, \dot{q}_{s}, q_{s}^{(3)})]$$
(9)

$$\alpha(q_{s}, \dot{q}_{s}, \ddot{q}_{s}, q_{s}^{(3)}) = 2\dot{M}_{s}(q_{s}, \dot{q}_{s})q_{s}^{(3)} + (\ddot{M}_{s}(q_{s}, \dot{q}_{s}, \ddot{q}_{s}) + K_{s})\ddot{q}_{s} + \ddot{N}(q_{s}, \dot{q}_{s}, \ddot{q}_{s}, q_{s}^{(3)})$$
(10)

From (1) it follows that  $\ddot{q}_s$ ,  $q_s^{(3)}$  are related to lower order variables  $q_s$ ,  $\dot{q}_s$ ,  $\dot{\theta}_s$  as:

$$\ddot{q}_{s} = -M_{s}^{-1}(q_{s})(N(q_{s},\dot{q}_{s}) + K_{s}(q_{s} - \theta_{s}))$$
(11)

$$q_{s}^{(5)} = -M_{s}^{-1}(q_{s})(M_{s}(q_{s},\dot{q}_{s})\ddot{q}_{s} + N(q_{s},\dot{q}_{s},\ddot{q}_{s}) + K_{s}(\dot{q}_{s} - \theta_{s}))$$
(12)

Therefore (9) can be written as function of  $q_s$ ,  $\dot{q}_s$ ,  $\theta_s$ ,  $\dot{\theta}_s$ , i.e.:

$$\tau_{s} = K_{s}(\theta_{s} - q_{s}) + \beta(q_{s}, \dot{q}_{s}, \theta_{s}, \theta_{s}) + \varphi(q_{s})v(t)$$
  
$$\beta(q_{s}, \dot{q}_{s}, \theta_{s}, \dot{\theta}_{s}) = B_{v,s}\dot{\theta}_{s} + J_{s}K_{s}^{-1}\alpha(q_{s}, \dot{q}_{s}, \ddot{q}_{s}, q_{s}^{(3)}) \quad (13)$$
  
$$\varphi(q_{s}) = J_{s}K_{s}^{-1}M_{s}(q_{s})$$

Applying the controller (9) to the system (1)–(3) yields the linear decoupled closed loop system:

$$q_s^{(4)} = v(t)$$
 (14)

To ensure synchronization between the slave and the master robot v(t) is proposed as:

$$p(t) = q_m^{(4)} - K_3 e^{(3)} - K_2 \ddot{e} - K_1 \dot{e} - K_0 e$$
(15)

with  $K_i \in \mathbb{R}^n$  gain matrices and the synchronization errors defined by:

$$e = q_s - q_m, \quad \dot{e} = \dot{q}_s - \dot{q}_m \tag{16}$$

where the master position trajectory  $q_m$  has to be at least four times differentiable, i.e.,  $q_m \in C^4$ . Clearly, there exist general choices for the gain matrices  $K_i$  such that the closed loop systems is stable, but for simplicity and without lost of generality, we assume that the gain matrix  $K_i$  is a multiple of the identity matrix, i.e.,  $K_i = k_i I_n$ , i = 0, 1, 2, 3, with  $k_i$  positive scalars. Then, it is straightforward to conclude that the synchronization error e is exponentially stable, if the scalars  $k_i$ , i = 0, 1, 2, 3 are chosen such that the polynomial  $s^4 + k_3 s^3 + k_2 s^2 + k_1 s + k_0$  is Hurwitz.

#### 4 Feedback Controller Based on Estimated Variables

As stated in Section 1, it is assumed that only the master and slave link positions  $q_m$ ,  $q_s$  are measured, therefore  $\tau_s$  (9) and v(t) (15) can not be implemented.

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Let  $\hat{q}_s$ ,  $\hat{q}_s$ ,  $\hat{\theta}_s$ ,  $\hat{\theta}_s$  denote estimated values for  $q_s$ ,  $\dot{q}_s$ ,  $\theta_s$ ,  $\dot{\theta}_s$ , and v(t) given by (15) is implemented based on estimated synchronization errors. Assuming that  $\hat{q}_s$ ,  $\hat{q}_s$ ,  $\hat{\theta}_s$ ,  $\hat{\theta}_s$  are available, the controller (13) can be modified as:

$$\tau_{s} = K_{s}(\hat{\theta}_{s} - \hat{q}_{s}) + \beta(q_{s}, \hat{q}_{s}, \hat{\theta}_{s}, \hat{\theta}_{s}) + \varphi(q_{s})v(t)$$
$$\beta(q_{s}, \hat{q}_{s}, \hat{\theta}_{s}, \hat{\theta}_{s}) = B_{v,s}\hat{\theta}_{s} + J_{s}K_{s}^{-1}\alpha(q_{s}, \hat{q}_{s}, \hat{q}_{s}, \hat{q}_{s})$$
(17)

$$\alpha(q_{s},\hat{q}_{s},\hat{q}_{s},q_{s}^{(3)}) = 2\dot{M}_{s}(q_{s},\hat{q}_{s})q_{s}^{(3)} + (\dot{M}_{s}(q_{s},\hat{q}_{s},\hat{q}_{s}) + K_{s})\hat{q}_{s} + \ddot{N}(q_{s},\hat{q}_{s},\hat{q}_{s},\hat{q}_{s},\hat{q}_{s})$$

where according to (11), (12), estimates for  $\ddot{q}_s$ ,  $q_s^{(3)}$  are given by

$$\hat{\hat{q}}_{s} = -M_{s}^{-1}(q_{s})(N(q_{s},\hat{q}_{s}) + K_{s}(\hat{q}_{s} - \hat{\theta}_{s}))$$
(18)

$$\tilde{q}_{s}^{(3)} = -M_{s}^{-1}(q_{s})(\dot{M}_{s}(q_{s},\hat{q}_{s})\hat{q}_{s} + \dot{N}(q_{s},\hat{q}_{s},\hat{q}_{s}) + K_{s}(\hat{q}_{s}-\hat{\theta}_{s}))$$
(19)

**4.1** An Observer for the Synchronization Errors. From the work of Berghuis and Nijmeijer [19], we propose the modified controller v(t) (15) as:

$$v = -K_3 \dot{w}_2 - K_2 \dot{w}_1 - K_1 \dot{\hat{e}} - K_0 \hat{e}$$
(20)

where  $\hat{e}$ ,  $\hat{e}$ ,  $\dot{w}_1$ ,  $\dot{w}_2$  represent estimates for e,  $\dot{e}$ ,  $\ddot{e}$ ,  $e^{(3)}$  respectively. They are obtained by the observer:

$$\hat{e} = w_1 + \Gamma_1 \tilde{e}$$
  

$$\dot{w}_1 = w_2 + \Gamma_2 \tilde{e}$$
  

$$\dot{w}_2 = w_3 + \Gamma_3 \tilde{e}$$
  

$$\dot{w}_3 = \Gamma_4 \tilde{e}$$
(21)

with  $\Gamma_i \in \mathbb{R}^{n \times n}$ , i = 1, 2, 3, 4 diagonal positive definite gain matrices and the estimation synchronization error defined by:

$$\tilde{e} = e - \hat{e} \tag{22}$$

**4.2** An Observer for the Slave Variables.  $q_s$ ,  $\dot{q}_s$ ,  $\theta_s$ ,  $\dot{\theta}_s$ . Based on the dynamic model (1)–(3), we propose the nonlinear Luenberger observer:

$$\frac{d}{dt}\hat{q}_{s} = \hat{q}_{s} + \mu_{1}\tilde{q}$$

$$\frac{d}{dt}\hat{q}_{s} = -M_{s}^{-1}(q_{s})(N(q_{s},\hat{q}_{s}) + K_{s}(\hat{q}_{s} - \hat{\theta}_{s})) + \mu_{2}\tilde{q}$$

$$\frac{d}{dt}\hat{q}_{s} = \hat{q}_{s} + \mu_{s}\tilde{q}$$
(23)

$$\frac{d}{dt} \theta_s = \theta_s + \mu_3 \tilde{q}$$

$$\frac{d}{dt} \hat{\theta}_s = J_s^{-1} (\tau_s(q_s, \hat{q}_s, \hat{q}_s, \hat{\theta}_s, \hat{\theta}_s) - K_s(\hat{\theta}_s - \hat{q}_s) - B_{\mu_s} \hat{\theta}_s) + \mu_4 \tilde{q}$$
(24)

 $\frac{d}{dt}\hat{\theta}_s = J_s^{-1}(\tau_s(q_s, \hat{q}_s, \hat{q}_s, \hat{\theta}_s, \hat{\theta}_s) - K_s(\hat{\theta}_s - \hat{q}_s) - B_{v,s}\hat{\theta}_s) + \mu_4 \tilde{q}$ 

where  $\hat{q}_s$ ,  $\hat{q}_s$ ,  $\hat{\theta}_s$ ,  $\hat{\theta}_s$  correspond to the estimates of  $q_s$ ,  $\dot{q}_s$ ,  $\theta_s$ ,  $\dot{\theta}_s$ , and  $\mu_1$ ,  $\mu_2$ ,  $\mu_3$ ,  $\mu_4 \in \mathbb{R}^{n \times n}$  are diagonal positive definite gain matrices. The joint estimation errors are defined by

$$\widetilde{q} = q_s - \hat{q}_s, \quad \widetilde{\theta} = \theta_s - \hat{\theta}_s \tag{25}$$

# 5 Closed-Loop System

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In this section, the total error dynamics is obtained.

**Proposition 1.** Consider the master and slave flexible joint robots, which are described by (1)-(3), the slave robot in closed loop with the synchronization controller  $\tau_s$  (17), v (20) and the observers (21), (23), and (24), then the synchronization closed-loop error dynamics is given by:

$$\begin{aligned} & (^{(4)} + K_3 e^{(3)} + K_2 \ddot{e} + K_1 \dot{e} + K_0 e \\ &= q_m^{(4)} + K_3 (\tilde{e}^{(3)} + \Gamma_1 \ddot{e} + \Gamma_2 \dot{e}) + K_2 (\ddot{e} + \Gamma_1 \dot{e}) + K_1 \dot{e} + K_0 \tilde{e} \\ &- M_s^{-1} (q_s) \Phi(q_s, \dot{q}_m, \ddot{q}_m, q_m^{(3)}, \dot{e}, \ddot{e}, e^{(3)}, \tilde{q}, \dot{\tilde{q}}, \tilde{\theta}, \dot{\tilde{\theta}}) \end{aligned} (26) \\ & \tilde{e}^{(4)} + (\Gamma_1 - K_3) \tilde{e}^{(3)} + (\Gamma_2 - K_3 \Gamma_1 - K_2) \ddot{\tilde{e}} \\ &+ (\Gamma_3 - K_3 \Gamma_2 - K_2 \Gamma_1 - K_1) \dot{\tilde{e}} + (\Gamma_4 - K_0) \tilde{e} \\ &= q_m^{(4)} - K_3 e^{(3)} - K_2 \ddot{e} - K_1 \dot{e} - K_0 e \\ &- M_s^{-1} (q_s) \Phi(q_s, \dot{q}_m, \ddot{q}_m, q_m^{(3)}, \dot{e}, \ddot{e}, e^{(3)}, \tilde{q}, \dot{\tilde{q}}, \tilde{\theta}, \dot{\tilde{\theta}}), \end{aligned} (27)$$

$$\begin{split} \ddot{\tilde{q}} &= -M_s^{-1}(q_s)((2C_s(q_s,\dot{e}+\dot{q}_m)-C_s(q_s,\ddot{q}+\mu_1\tilde{q}))(\dot{\tilde{q}}+\mu_1\tilde{q}) \\ &+K_s(\tilde{q}-\tilde{\theta}))-\mu_1\dot{\tilde{q}}-\mu_2\tilde{q} \\ &\ddot{\tilde{\theta}} &= -J_s^{-1}K_s(\tilde{\theta}-\tilde{q})-J_s^{-1}B_{v,s}(\dot{\tilde{\theta}}+\mu_3\tilde{q})-\mu_3\dot{\tilde{q}}-\mu_4\tilde{q} \end{split}$$
(28)

**Proof:** A sketch of the proof is given in Appendix 1. Notice that  $\Phi$  can be considered as a disturbance term that affects (26) and (27). Also notice that (26) and (27) present a linear structure, therefore, the following proposition can be formulated.

**Proposition 2.** Define states  $x_1, \ldots, x_8 \in \mathbb{R}^n$  as  $x_1 = e, x_2 = \dot{e}, x_3 = \ddot{e}, x_4 = e^{(3)}, x_5 = \tilde{e}, x_6 = \dot{e}, x_7 = \ddot{e}, x_8 = \tilde{e}^{(3)}, and y \in \mathbb{R}^{4n}$  as  $y_1 = \tilde{q}, y_2 = \ddot{q}, y_3 = \tilde{\theta}, y_4 = \dot{\theta}$ . Such that the state vectors x and y are given by  $x = [x_1^T x_2^T x_3^T x_4^T x_5^T x_6^T x_7^T x_8^T]^T$  and  $y = [y_1^T y_2^T y_3^T y_4^T]^T$ . Then (26), (27), and (28) can be written as:

$$\dot{x} = Ax - BM_s^{-1}(q_s)\Phi(q_s, \dot{q}_m, \ddot{q}_m, q_m^{(3)}, x_2, x_3, x_4, y_1, y_2, y_3, y_4) + Bq_m^{(4)}$$

$$\dot{y} = f(q_s, \dot{q}_m, x, y) \tag{29}$$

$$A = \begin{bmatrix} 0 & I_n & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & I_n & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & I_n & 0 & 0 & 0 & 0 \\ -K_0 & -K_1 & -K_2 & -K_3 & K_0 & \pi_1 & \pi_2 & K_3 \\ 0 & 0 & 0 & 0 & I_n & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & I_n & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & I_n & 0 \\ -K_0 & -K_1 & -K_2 & -K_3 & K_0 - \Gamma_4 & \pi_1 - \Gamma_3 & \pi_2 - \Gamma_2 & K_3 - \Gamma_1 \end{bmatrix} \in \mathbb{R}^{8n \times 8n}$$
(30)  
$$B = \begin{bmatrix} 0 & 0 & 0 & I_n & 0 & 0 & I_n \\ -K_0 & -K_1 & -K_2 & -K_3 & K_0 - \Gamma_4 & \pi_1 - \Gamma_3 & \pi_2 - \Gamma_2 & K_3 - \Gamma_1 \end{bmatrix}$$

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(31)

$$\pi_1 = K_3 \Gamma_2 + K_2 \Gamma_1 + K_1, \quad \pi_2 = K_3 \Gamma_1 + K_2$$

with 0,  $I_n \in \mathbb{R}^{n \times n}$  the zero and identity matrices.

The nonlinear vector function  $f(q_s, \dot{q}_m, x, y)$  is given by:

$$f(q_{s},\dot{q}_{m},x,y) = \begin{bmatrix} y_{2} \\ -M_{s}^{-1}(q_{s})(2C_{s}(q_{s},x_{2}+\dot{q}_{m})-C_{s}(q_{s},y_{2}+\mu_{1}y_{1}))(y_{2}+\mu_{1}y_{1}) \\ -M_{s}^{-1}(q_{s})K_{s}(y_{1}-y_{3})-\mu_{1}y_{2}-\mu_{2}y_{1} \\ y_{4} \\ -J_{s}^{-1}K_{s}(y_{3}-y_{1})-J_{s}^{-1}B_{v,s}(y_{4}+\mu_{3}y_{1})-\mu_{3}y_{2}-\mu_{4}y_{1} \end{bmatrix} \in \mathbb{R}^{4n}$$
(32)

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*Proof*: This follows from simple substitution of the states x and y.

# 6 Stability Analysis

The stability analysis is based on a Lyapunov function, whose derivative can be bounded in terms of the closed-loop errors. It is proven that the bound is negative in an annulus around the origin. To derive bounds on the derivative of the Lyapunov function the following assumption is required.

**Assumption 3.** The signals  $\dot{q}_m$ ,  $\ddot{q}_m$ ,  $q_m^{(3)}$ ,  $q_m^{(4)}$  are bounded for all  $t \in [t_0, \infty)$ , therefore there exist  $V_M$ ,  $A_M$ ,  $D_M$  and  $E_M$  such that:

$$\sup_{t} \|\dot{q}_{m}\| = V_{M} < \infty, \quad \sup_{t} \|\ddot{q}_{m}\| = A_{M} < \infty$$

$$\sup_{t} \|q_{m}^{(3)}\| = D_{M} < \infty, \quad \sup_{t} \|q_{m}^{(4)}\| = E_{M} < \infty$$
(33)

In practice, it is often not difficult to obtain the master trajectories bounds (33) on the basis of the master desired trajectories  $q_d(t)$ and its derivatives, although due to friction effects, tracking errors, etc., the actual motion of the master robot may differ from its desired motion. Also the bounds  $V_M$ ,  $A_M$ ,  $D_M$ , and  $E_M$  can be obtained by considering the structural limitations of the robots, such as maximum velocities and accelerations of the motors. On the other hand, for the sake of simplicity the following assumption is imposed.

**Assumption 4.** All the gains in the controller (17), (20), and the observer (21) are a positive multiple of the unit matrix, i.e., of the form K = kI, where k is a positive scalar.

Based on the above assumptions, the main result of this paper is formulated as follows.

**Theorem 5.** Consider the master and slave flexible joint robots, described by (1)-(3), the slave robot in closed loop with the controller  $\tau_s$  (17), and v (20) and the observers (21), (23) and (24). Assume that the gain matrices  $K_i$  and  $\Gamma_j$ , i=0, 1, 2, 3, j=1, 2, 3, 4 are chosen such that the matrix A, given by (30), is Hurwitz, and additionally the minimum and maximum eigenvalues of the gains  $\mu_l$ , l=1, 2, 3, 4, i.e.,  $\mu_{lm}$  and  $\mu_{lM}$ , satisfy:

$$\mu_{1m} > \max\{0, 2M_{sm}^{-1}(\lambda_0 - C_M V_M), (2\lambda_0 M_{sm})^{-1} \\ \times (4\lambda_0^2 - M_{sm} K_{sm} - 2C_M V_M (2\lambda_0 + M_{sm} \mu_{1M}) \\ + M_{sm} M_{sM} \mu_{2M})\}$$
(34)

$$\mu_{2m} > \max\{0, -a_{31}^{-1}a_{30}, -a_{41}^{-1}a_{40}\}$$
(35)

$$\mu_{3m} > \max\{0, \mu_{33}, \mu_{34}\}, \quad \mu_{3M} < \min\{\mu_{33}, \mu_{34}\}$$
(36)

$$\mu_{4m} > \max\{0, J_{sm}^{-1}(4\eta_0^2 - J_{sm}K_{sm} - 2\eta_0 B_{v,sM} + J_{sm}\mu_{4M})\}$$
(37)

where  $M_{sm}$  and  $J_{sm}$  are the minimum eigenvalue of the link and motor rotor inertia matrices, and the scalars  $\lambda_0$ ,  $\eta_0$ ,  $\mu_{33}$ ,  $\mu_{34}$ ,  $\mu_{33}^-$ ,  $\mu_{34}^-$ ,  $a_{30}^-$ ,  $a_{31}$ ,  $a_{40}$ ,  $a_{41}$  are defined via the gain tuning procedure of Section 7. Then the synchronization closed-loop errors and the estimation errors are semi-globally uniformly ultimately bounded.

In particular, it means that ther exist a region of convergence depending on the controller gains, such that if the initial errors at time t=0 are in this region they will remain bounded for all time  $t \ge 0$  with a bound smaller that the region of convergence.

*Proof*: The proof is divided in three parts, first the candidate Lyapunov function and conditions for positive definitiveness are presented. Second the derivative of the Lyapunov function along (29) is bounded, and finally sufficient conditions for negative definiteness are formulated.

**6.1 Lyapunov Function.** Consider the synchronization closed-loop error dynamics given by (29), and take as a candidate Lyapunov function:

$$V(q_{s},x,y) = x^{T} P_{x} x + \frac{1}{2} y^{T} P_{y}(q_{s},y) y$$
(38)

where the positive definite symmetric matrix  $P_x$  is the solution of the Lyapunov equation  $P_x A + A^T P_x = -Q_x$ , for any given symmetric positive definite matrix  $Q_x$ , and  $P_y(q_x, y)$  is given by:

$$P_{y}(q_{s},y) = \begin{bmatrix} K_{s} + 2\lambda_{0}\mu_{1} + \beta_{1}I_{n} & 2\lambda_{0}I_{n} \\ 2\lambda_{0}I_{n} & M_{s}(q_{s}) \end{bmatrix} & 0 \\ 0 & \begin{bmatrix} K_{s} + \mu_{4} + \beta_{2}I_{n} & 2\eta(y_{3})I_{n} \\ 2\eta(y_{3})I_{n} & J_{s} \end{bmatrix} \end{bmatrix}$$

T

with  $\beta_1$ ,  $\beta_2$  scalars to be determined, and  $\eta(y_3)$  defined by:

 $\lambda_0$ ,  $\eta_0 > 0$  are constant scalars; then for all  $\tilde{\theta}$ ,  $\tilde{\theta}$  it holds that:

$$0 < \|\eta(y_3)\| < \eta_0, \quad \dot{\eta}(y_3) y_4^T y_3 \le \eta_0 \|y_4\|^2$$

Existence of  $P_x$  is guaranteed if A, which is given by (30), is

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 $\eta(y_3) \coloneqq \frac{\eta_0}{1 + \|y_3\|}$ 

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Hurwitz. By Assumption 4, a sufficient condition for A being Hurwitz is that the scalar polynomial  $\lambda(s)$ , given by:

0

$$\lambda(s) = s^{8} + \gamma_{1}s^{7} + \gamma_{2}s^{6} + \gamma_{3}s^{5} + (\gamma_{4} + k_{1}\gamma_{1} + k_{2}\gamma_{2} + k_{3}\gamma_{3})s^{4} + + (k_{0}\gamma_{1} + k_{1}\gamma_{2} + k_{2}\gamma_{3} + k_{3}\gamma_{4})s^{3} + (k_{0}\gamma_{2} + k_{1}\gamma_{3} + k_{2}\gamma_{4})s^{2} + (k_{0}\gamma_{3} + k_{1}\gamma_{4})s + k_{0}\gamma_{4}$$
(39)

is Hurwitz.  $k_i$ ,  $\gamma_j$  are the scalars associated to the gains  $K_i$  and  $\Gamma_j$ , i=0, 1, 2, 3, j=1, 2, 3, 4. Thus by choosing the gains  $K_i$  and  $\Gamma_j$ , it can be ensured that the matrix A is Hurwitz. Therefore, there exists a unique positive symmetric matrix  $P_x$ , which satisfies  $P_x A + A^T P_x = -Q_x$ , for any given symmetric positive definite matrix  $Q_x$ .

On the other hand, sufficient conditions for positive definiteness of  $P_{y}(q_{s}, y)$  are:

$$\mu_{1m} > \frac{1}{2\lambda_0 M_{sm}} (4\lambda_0^2 - M_{sm} K_{sm} - M_{sm} \beta_1)$$

$$\mu_{4m} > \frac{1}{J_{sm}} (4\eta_0^2 - J_{sm} K_{sm} - J_{sm} \beta_2)$$
(40)

Finally, positive definiteness of  $P_x$  and  $P_y(q_s, y)$  imply positive definiteness of  $V(q_s, x, y)$  (38). Moreover, there exist positive with  $Q_x \in \mathbb{R}^{8n \times 8n}$  the symmetric positive definite matrix  $Q_x = -(P_x A + A^T P_x)$  and  $Q_y \in \mathbb{R}^{4n \times 4n}$ 

$$\frac{1}{2} P_m \|\xi\|^2 \leq V(q_s, x, y) \leq \frac{1}{2} P_M \|\xi\|^2$$
(41)

6.2 Derivative of the Lyapunov Function. Along the error dynamics (29), the time derivative of V (38) is given by:

$$\dot{V} = -x^{T}Q_{x}x - x^{T}P_{x}BM_{s}^{-1}\Phi - \Phi^{T}M_{s}^{-1}B^{T}P_{x}x$$

$$+y^{T}P_{y}(q_{s}, y)f(x, y) + \frac{1}{2}y^{T}\dot{P}_{y}(q_{s}, y)y + x^{T}P_{x}Bq_{m}^{(4)}$$

$$+q_{m}^{(4)T}B^{T}P_{x}x \qquad (42)$$

that can be written as

$$\dot{V} = -\begin{bmatrix} x & y \end{bmatrix}^{T} \begin{bmatrix} Q_{x} & Q_{xy} \\ Q_{xy}^{T} & Q_{y} \end{bmatrix} \begin{bmatrix} x \\ y \end{bmatrix} + \Omega(q_{x}, \dot{q}_{m}, \ddot{q}_{m}, q_{m}^{(3)}, q_{m}^{(4)}, x, y)$$
(43)

$$Q_{y} = \begin{bmatrix} 2\lambda_{0}(M_{s}^{-1}K_{s} + \mu_{2}) & \alpha_{1} & \alpha_{2} & \alpha_{3} \\ \alpha_{1}^{T} & M_{s}\mu_{1} - 2\lambda_{0} & \eta(y_{3})\mu_{3} - \frac{1}{2}K_{s} & \frac{1}{2}J_{s}\mu_{3} \\ \alpha_{2}^{T} & \left(\eta(y_{3})\mu_{3} - \frac{1}{2}K_{s}\right)^{T} & 2\eta(y_{3})J_{s}^{-1}K_{s} & \alpha_{4} \\ \alpha_{3}^{T} & \frac{1}{2}(J_{s}\mu_{3})^{T} & \alpha_{4}^{T} & B_{v,s} - 2\eta(y_{3})I_{n} \end{bmatrix} \\ \alpha_{1} = -\frac{1}{2}(\beta_{1}I_{n} + M_{s}\mu_{2}) \\ \alpha_{2} = -\lambda_{0}M_{s}^{-1}K_{s} + \eta(y_{3})(\mu_{4} + J_{s}^{-1}B_{v,s}\mu_{3} - J_{s}^{-1}K_{s}) \\ \alpha_{3} = \frac{1}{2}(B_{v,s}\mu_{3} + J_{s}\mu_{4} - K_{s}) \\ \alpha_{4} = \eta(y_{3})J_{s}^{-1}B_{v,s} - \frac{1}{2}(\mu_{4} + \beta_{2}I_{n}) \end{cases}$$

$$(44)$$

 $Q_{xy} \in \mathbb{R}^{8n \times 4n}$  depends on the entries of  $P_x \in \mathbb{R}^{8n \times 8n}$ , and it is given by:

$$Q_{xy} = \begin{bmatrix} P_{x14} + P_{x18} \\ P_{x24} + P_{x28} \\ P_{x34} + P_{x38} \\ P_{x44} + P_{x48} \\ P_{x45}^T + P_{x58} \\ P_{x46}^T + P_{x68} \\ P_{x46}^T + P_{x78} \\ P_{x48}^T + P_{x88} \end{bmatrix} M_s^{-1} K_s J_s^{-1} [(B_{v,s}\mu_3 - K_s) \ 0 \ K_s \ B_{v,s}]$$

$$(45)$$

The scalar function  $\Omega(q_s, \dot{q}_m, \ddot{q}_m, q_m^{(3)}, q_m^{(4)}, x, y)$  is locally Lipschitz and is given by:

$$\Omega = -x^{T} P_{x} B M_{s}^{-1} (\Phi + K_{s} J_{s}^{-1} K_{s} (y_{1} - y_{3}) - K_{s} J_{s}^{-1} B_{v,s} (y_{4} + \mu_{3} y_{1})) + x^{T} P_{x} B q_{m}^{(4)} + - (\Phi + K_{s} J_{s}^{-1} K_{s} (y_{1} - y_{3}) - K_{s} J_{s}^{-1} B_{v,s} (y_{4} + \mu_{3} y_{1}))^{T} M_{s}^{-1} B^{T} P_{x} x + q_{m}^{(4)T} B^{T} P_{x} x - (2\lambda_{0} y_{1}^{T} M_{s}^{-1} + y_{2}^{T}) (2C_{s} (q_{s}, x_{2} + \dot{q}_{m}) - C_{s} (q_{s}, \mu_{1} y_{1})) (y_{2} + \mu_{1} y_{1}) + 2\lambda_{0} y_{1}^{T} M_{s}^{-1} C_{s} (q_{s}, y_{2}) (y_{2} + \mu_{1} y_{1}) + y_{2}^{T} C_{s} (q_{s}, y_{2}) \mu_{1} y_{1} + 2\dot{\eta} (y_{3}) y_{4}^{T} y_{3}$$
(46)

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6.2.1 Boundedness of  $\dot{V}$ . The next lemma synthesizes a general bound for  $\dot{V}$  given by (43).

**Lemma 6.** Consider  $\dot{V}$  given by (43) and introduce the vectors  $x_N, y_N$ , and  $\xi_N$ :

$$x_{N} = [\|x_{1}\| \|x_{2}\| \|x_{3}\| \|x_{4}\| \|x_{5}\| \|x_{6}\| \|x_{7}\| \|x_{8}\|]^{T}$$

$$y_{N} = [\|y_{1}\| \|y_{2}\| \|y_{3}\| \|y_{4}\|]^{T} \qquad (47)$$

$$\xi_{N} = [x_{N}^{T} \ y_{N}^{T}]^{T}$$

Then V given by (43) can be upper-bounded as:

$$V \leq -\xi_N^T R_V \xi_N + \Theta(V_M, A_M, D_M, E_M, \xi_N)$$
(48)

with  $V_m$ ,  $A_m$ ,  $D_m$ ,  $E_m$  given by (33),  $\Theta(V_m, A_m, D_m, E_m, \xi_N)$  a scalar function that contains products of at most order 5 in terms of the entries of  $\xi_N$ , and the matrix  $R_V$  is given by

$$R_{V} = \begin{bmatrix} R_{x} & R_{xy}(V_{M}, A_{M}, D_{M}) \\ R_{xy}^{T}(V_{M}, A_{M}, D_{M}) & R_{y}(V_{m}) \end{bmatrix}$$
(49)

 $R_x$ ,  $R_{xy} \in \mathbb{R}^{8 \times 4}$ ,  $R_y \in \mathbb{R}^{4 \times 4}$ ,  $\Theta$  and the positive definite matrix  $R_x \in \mathbb{R}^{8 \times 8}$  as in Appendix 2.

Proof: See Appendix 2.

The upperbound of V given by (48) can be reduced to a function of the norm of  $\xi_N$ .

**Lemma 7**. In terms of the vector  $\xi_N$ , (47), the upper-bound of  $\dot{V}$ , (48) can be reduced to

$$\dot{V} \leq \|\xi_{N}\|(r_{0} - r_{1}\|\xi_{N}\| + r_{2}\|\xi_{N}\|^{2} + r_{3}\|\xi_{N}\|^{3} + r_{4}\|\xi_{N}\|^{4})$$
(50)

where  $r_1$  is the minimum eigenvalue of the matrix  $R_V$ , (49), and the positive scalars  $r_0(E_M)$ ,  $r_2(\mu_{1M}, \mu_{3M})$ ,  $r_3(\mu_{1M})$ ,  $r_4(\mu_{1M})$ are determined by  $\Theta(V_M, A_M, D_M, E_M, \xi_N)$ , with  $V_M, A_M, D_M$ ,  $E_M$  the upperbounds for  $\dot{q}_m$ ,  $\ddot{q}_m$ ,  $q_m^{(3)}$ ,  $q_m^{(4)}$ , which are defined by (33).

*Proof*: This follows directly from (48) and the definition of  $\xi_N$ .

6.2.2 Negative Definiteness of V. Equation (50) is an upperbound for V, such that V is negative if and only if  $r_1$  is positive and the vector  $\xi_N$  holds  $x_{\xi,1} < \|\xi_N\| < x_{\xi,2}$ . Where  $x_{\xi,1}$  and  $x_{\xi,2}$  are the roots of the polynomial:

$$g(x_{\xi}) = r_0 - r_1 x_{\xi} + r_2 x_{\xi}^2 + r_3 x_{\xi}^3 + r_4 x_{\xi}^4$$
(51)

such that  $g(x_{\xi}) \leq 0$  if  $x_{\xi,1} \leq x_{\xi} \leq x_{\xi,2}$ .

Notice that  $x_{\xi,1}$  and  $x_{\xi,2}$  determine the region in which  $\dot{V}$  is negative definite. Thus  $x_{\xi,1}$  and  $x_{\xi,2}$  together with the minimum  $P_m$  and maximum  $P_M$  bounds of the Lyapunov function V, (41), determine the region of convergence of the synchronization closed loop system, similar as in [16]. The scalar  $r_1$  is the minimum eigenvalue of the matrix  $R_V$  (49), so,  $r_1$  is positive if and only if  $R_V$  is positive definite. The following lemma is useful to prove positive definiteness of  $R_V$ .

**Lemma 8.** (See [20]) If  $L \in \mathbb{R}^{m \times m}$  and  $M \in \mathbb{R}^{n \times n}$  are given positive semidefinite matrices and  $X \in \mathbb{R}^{m \times n}$ , then the symmetric block matrix:

$$\chi_{LM} = \begin{bmatrix} L & X \\ X^T & M \end{bmatrix}$$
(52)

is positive semidefinite if and only if there exists a matrix  $C \in \mathbb{R}^{m \times n}$ , such that  $X = L^{1/2}CM^{1/2}$ . If L and M are positive definite, then this criterion is equivalent to:

$$\|L^{-1/2}XM^{-1/2}\|_2 \leq 1$$

Moreover  $\chi_{LM}$  is positive definite if and only if L and M are positive definite and

$$\|L^{-1/2}XM^{-1/2}\|_2 < 1 \tag{53}$$

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Before applying the above lemma, it is required to prove that the matrix  $R_v$  is positive definite.

**Proposition 9.** Consider  $R_y \in \mathbb{R}^{4 \times 4}$  the matrix defined in (49), and define scalars  $\beta_1$ ,  $\beta_2$  as:

$$\beta_1 = 2C_M V_M (2M_{sm}^{-1}\lambda_0 + \mu_{1M}) - M_{sM}\mu_{2M}$$

$$\beta_2 = 2\eta_0 J_{sm}^{-1} B_{v,sM} - \mu_{4M}$$
(54)

Then sufficient conditions for positive definiteness of  $R_{y}$  are:

1. 
$$\Lambda_0 > 0$$
  
2.  $\mu_{1m} > \max\{0, 2M_{sm}^{-1}(\lambda_0 - C_M V_M)\}$ 

3. 
$$0 < \eta_0 < \min\{B_{v,sm}/4, \frac{1}{8J_{sM}}(J_{sM}B_{v,sm}/4, 1) + 8J_{sM}J_{cm}^2K_{sm}^{-1/2}\}$$

4. 
$$\mu_{3m} > \max\{0, \mu_{33}, \mu_{34}, \mu_{3M} < \min\{\mu_{33}, \mu_{34}\}, \mu_{3M} < \min\{\mu_{33}, \mu_{34}\}$$

$$\mu_{2m} > \max\{0, -a_{31}^{-1}a_{30}, -a_{41}^{-1}a_{40}\}$$

with  $\mu_{33_-}$ ,  $\mu_{34_-}$ ,  $\mu_{33}^-$ ,  $\mu_{34}^-$ ,  $a_{30}$ ,  $a_{31}$ ,  $a_{40}$ ,  $a_{41}$  given in Appendix 3.

Proof: See Appendix 3.

The conditions listed in Theorem 5 clearly imply the conditions in the above lemma, therefore it can be ensured that  $R_y$  is positive definite and then Lemma 8 can be used.

For  $R_V$ , given by (49), the condition (53) can be written as:

$$\|R_x^{-1/2}R_{xy}R_y^{-1/2}\|_2 < 1$$

Notice that  $R_{xy}$  depends on the gains  $\mu_1$  and  $\mu_3$ , but does not depend on  $\mu_2$ ,  $\eta_0$ ,  $\lambda_0$ . Then, if the gains  $\mu_1$ ,  $\mu_3$  have been chosen according to Proposition 9, it follows that  $R_{xy}$  is only determined by  $P_x$ , (38), that is determined by  $Q_x$  (43). The matrix  $R_x$  in  $R_V$ , given by (49) is only determine by the positive definite symmetric matrix  $Q_x$ , therefore by choosing  $Q_x$  it can be ensured that det $(Q_x) \ge 1$  and thus det $(R_x) \ge 1$ . Also notice that the only entry of  $R_v$  which depends on  $\mu_2$  is  $\alpha_0^*$  thus by choosing the minimum eigenvalue of the gain  $\mu_2$  it can be ensured that  $\det(R_y) \ge 1$ . There-fore, it follows that the entries of  $R_x^{-1/2} R_{xy} R_y^{-1/2}$  are small. As a result  $||R_x^{-1/2}R_{xy}R_y^{-1/2}||_2 < 1$  can be ensured, and thus  $R_V$ , given by (49), is positive definite. Since the condition for  $R_V$  being positive definite is given by the minimum eigenvalue of the gain  $\mu_2$  then it follows that the minimum eigenvalue of  $R_V$ , i.e.,  $r_1$ , is determined by the minimum eigenvalue of  $\mu_2$ , that implies that  $r_1$  can be chosen such that it dominates the other terms in (50). To emphasize the last conclusion, notice that  $r_0$ ,  $r_2$ ,  $r_3$ ,  $r_4$  do no depend on the gain  $\mu_2$ .

Finally, notice that according to Section 6.1, Proposition 9, and the above paragraph, it follows that if conditions in Theorem 5 are fulfilled, then the function V, given by (38), is a Lyapunov function with  $\dot{V} < 0$  in an annulus around the origin determine by  $x_{\xi,1}$ ,  $x_{\xi,2}$ , (51), and the minimum  $P_m$  and maximum  $P_M$  bounds of the Lyapunov function V, (41), see [16]. Therefore the synchronization closed loop errors are uniformly ultimately bounded in such annulus.

The ultimate boundedness result is due to the absence of measurements of derivatives of the master trajectory  $q_m$ , therefore, we have the following corollary.

**Corollary 10.** If set point regulation of the master robot is considered and the master robot controller is able to achieve steady state in finite time, then  $q_m^{(4)}(t) = 0$  for  $t \in (t_2, \infty)$ , with  $t_2 \ge t_0$ , the convergence time of the master robot trajectories. If additionally the conditions on Theorem 5 are satisfied, then the controller (17), (20), and the observers (21), (23), and (24) yield local exponential convergence of the closed-loop errors.

*Proof*: If  $q_m^{(4)}(t) = 0$  for  $t \in (t_2, \infty)$ ,  $t_2 \ge t_0$ , with  $t_2$  the time in which the master robot achieves stationary state, then it implies that the upper-bound for  $q_m^{(4)}$  is zero for  $t \in (t_2, \infty)$ ,  $t_2 \ge t_0$ , and thus from Assumption 3 it follows that:

$$E_M = \sup_{t \ge t_2} ||q_m^{(4)}|| = 0$$



Fig. 1 Master and slave link positions  $q_m$ ,  $q_s$ , and synchronization position error e

so  $r_0(E_M) = 0$  in (50). Thus, if conditions in Theorem 5 are satisfied, for  $t \ge t_2$  (50) reduces to:

$$\dot{V} \leq \|\xi_N\|^2 (-r_1 + r_2 \|\xi_N\| + r_3 \|\xi_N\|^2 + r_4 \|\xi_N\|^3)$$
(55)

with  $r_1$ ,  $r_2$ ,  $r_3$ ,  $r_4>0$ , and  $\dot{V} \le 0$  if  $x_{\xi,1} < \|\xi_N\| < x_{\xi,2}$ . As a consequence, there exist a positive scalar  $\kappa$ , such that  $\dot{V}$  can be upperbounded as:

$$\dot{V} \le -\kappa \|\xi_N\|^2$$
 for all  $t \ge t_2$ ,  $x_{\xi,1} < \|\xi_N\| < x_{\xi,2}$ 

From the last equation and (41), we conclude that there exist some constants  $m^*$ ,  $\rho > 0$ , such that:

$$\|\xi_N(t)\|^2 \le m^* e^{-\rho t} \|\xi_N(t_2)\|^2$$
 for all  $t \ge t_2, x_{\xi, 1} < \|\xi_N\| < x_{\xi, 2}$ 

and thus from  $\xi_N(t)$ , (47), it follows that the close-loop errors are locally exponentially stable.

**Remark 11**. The proposed synchronization controller (17,20) is designed to guarantee synchronization between two robots. Nevertheless, it can be used as a tracking controller by taking the desired trajectory  $q_d(t)$  as the master robot trajectory  $q_m(t)$ . In case of tracking the desired trajectory  $q_d(t)$  and its derivatives are known, such that  $q_m^{(4)} = q_d^{(4)}$  can be included through the control v(t) (20). In such case, the closed-loop error (29) does not depend on  $q_m^{(4)} = q_d^{(4)}$ , and the stability analysis would result in  $\dot{V}$ given by (55). Therefore, for tracking of a known desired reference  $q_d(t)$  the proposed synchronization controller (17), (20) with  $\hat{v}(t)$ (20) modified as:

$$\hat{v}(t) = q_d^{(4)}(t) - K_3 \dot{w}_2 - K_2 \dot{w}_1 - K_1 \dot{\hat{e}} - K_0 \hat{e}$$

yields semi-global exponential convergence of the closed-loop errors.

#### 7 Design Procedure

The tuning gain procedure can be summarized as follows:

- 1. Choose the gains  $K_i$  and  $\Gamma_j$ , i=0, 1, 2, 3, j=1, 2, 3, 4 such that  $\lambda(s)$  (39) is Hurwitz.
- 2. Determine the bounds of the physical parameters  $M_s(q_s)$ ,  $C_s(q_s, \dot{q}_s)$ ,  $g_s(q_s)$  and their partial derivatives with respect to  $q_s$ .
- 3. Determine the bounds of the master trajectories  $\dot{q}_m$ ,  $\ddot{q}_m$ ,  $q_m^{(3)}$ ,  $q_m^{(4)}$ .
- 4. Choose  $\lambda_0 > 0$ ,  $\mu_{1M} > 0$ ,  $\mu_{4M} > 0$  and a bound for the maximum eigenvalue of  $\mu_2$ , i.e.,  $\mu_{2M}$ .

5. Choose 
$$\mu_1$$
, such that  $\mu_{1m} > \max\{0, 2M_{sm}^{-1}(\lambda_0 - C_M V_M), (2\lambda_0 M_{sm})^{-1}(4\lambda_0^2 - M_{sm} K_{sm} - 4C_M V_M \lambda_0 - 2C_M V_M M_{sm} \mu_{1M} + M_{sm} M_{sM} \mu_{2M})\}$ 

6. Choose  $\mu_0$ , such that  $0 < \mu_0 < \min\{B_{v,sm}/4, \frac{1}{8J_{sM}}(J_{sM}B_{v,sm} + (J_{sM}^2B_{v,sm}^2 + 8J_{sM}J_{sm}^2K_{sm})^{1/2})\}$ 7. Select  $\mu_3$ , such that  $\mu_{3m} > \max\{0, \mu_{33_-}, \mu_{34_-}\}, \mu_{3M}$ 

7. Select  $\mu_3$ , such that  $\mu_{3m} > \max\{0, \mu_{33}, \mu_{34}, \mu_{3M} \\ < \min\{\mu_{33}, \mu_{34}, \mu_{34}\}$ , see Appendix 3.

- 8.  $\mu_{4m} > \max\{0, J_{sm}^{-1}(4\eta_0^2 J_{sm}K_{sm} 2\eta_0B_{v,sM} + J_{sm}\mu_{4M})\}$
- Choose Q<sub>x</sub> a symmetric positive definite block diagonal matrix, with n×n block entries, such that det(Q<sub>x</sub>)≥1.
- 10. Determine  $P_x$  such that  $P_x A + A^T P_x = -Q_x$
- 11. Choose  $\mu_2$ , such that  $\mu_{2m} > \max\{0, -a_{31}^{-1}a_{30}, -a_{41}^{-1}a_{40}\}$ , see Appendix 3, and  $\mu_{2m}$  large enough to ensure  $\|R_x^{-1/2}R_{xy}R_y^{-1/2}\|_2 < 1$ .

#### 8 Simulations

The slave (s) and master (m) robot considered in the simulation consist of one rigid link with a flexible joint, rotating in a vertical plane. The dynamic model is given by:

$$M_i \ddot{q}_i + K_i (q_i - \theta_i) + \frac{1}{2} m_i g l_i \sin(q_i) = 0, \quad i = m, \quad s$$
$$J_i \ddot{\theta}_i + K_i (\theta_i - q_i) + B_{v,i} \dot{\theta}_i = \tau_i$$

The master robot parameters are  $M_m = 0.5$ ,  $K_m = 75$ ,  $B_{v,m} = 2$ ,  $m_m = 1.4$ ,  $l_m = 1$ ,  $J_m = 0.04$  (all values are in SI units), and its initial conditions are  $q_m(0) = 1$  rad,  $\dot{q}_m(0) = 0$  rad/s,  $\theta_m(0) = 1.1$  rad,  $\dot{\theta}_m(0) = 0$  rad/s. The master robot is driven by the controller (9) and (15), the gains on (15) are chosen as  $k_{0m} = 1$ ,  $k_{1m} = 3$ ,  $k_{2m} = 6$ ,  $k_{3m} = 3$ . The desired link master trajectory is:

$$q_{md}(t) = 1 + 0.5 \sin(t)$$
 [rad]

The slave robot parameters are  $M_s = 0.4$ ,  $K_s = 100$ ,  $B_{v,s} = 5$ ,  $m_s = 1$ ,  $l_s = 1$ ,  $J_s = 0.02$ . The initial conditions for the slave robot are  $q_s(0) = 0.5$  rad,  $\dot{q}_s(0) = 0$  rad/s,  $\theta_s(0) = 0.51$  rad,  $\dot{\theta}_s(0) = 0$  rad/s. The initial conditions for the observers (21), (23), and (24) are chosen as  $\hat{e}(0) = -0.1$ ,  $w_1(0) = 0$ ,  $w_2(0) = 0$ ,  $w_3(0) = 0$ ,  $\hat{q}_s(0) = 0.4$ ,  $\hat{q}_s(0) = 0.4$ , and  $\hat{\theta}_s(0) = 0$ .

The scalar gains in  $\tau_s$ , (20), and the observers (21), (23), and (24) are chosen to be  $k_0=65$ ,  $k_1=40$ ,  $k_2=10$ ,  $k_3=4$ ,  $\gamma_1=40$ ,  $\gamma_2=700$ ,  $\gamma_3=4000$ ,  $\gamma_4=1000$ ,  $\mu_1=1$ ,  $\mu_2=5$ ,  $\mu_3=1$ ,  $\mu_4=5$ . As it is shown in Fig. 1, the synchronization error between the master and slave link position is stable and bounded after the transient period has finished. The same is concluded for the estimation errors in Figs. 2 and 3.

In agreement with the stability analysis, the simulations have shown that the final bound of the errors depends on the gains  $\mu_2$ and  $K_0$ , see Section 6.2.2. Meanwhile the transient behavior is mainly determined by  $\mu_1$ ,  $\mu_3$  and the gain  $K_1$ , this is due to the



Fig. 2 Input torque  $\tau_s$  and estimation synchronization position error  $\tilde{e}$ 

fact the  $K_1$  defines the poles of the linear part Ax in (29), and  $\mu_1$ ,  $\mu_3$  weight the effect of the estimation errors through the term  $\Phi$  in (29). Therefore if  $\mu_1$ ,  $\mu_3$  are large, as well as the estimation errors, then the term  $\Phi$  has a large influence in the synchronization error dynamics, this large influence lasts until the estimation errors reach a vicinity around zero.

The simulation study shows that in order to minimize the peaks during the transient period, it is important to tune the gains on v(20) such that the polynomial  $\lambda(s)$  (39) corresponds to an overdamped system. At the same time the gains  $\mu_1$ ,  $\mu_3$  should be set small to minimize the influence (through the term  $\Phi$ ) of the estimation errors.

#### 9 Conclusions

A synchronization controller for flexible joint robots interconnected on a master-slave scheme, has been proposed. The controller only requires measurements of the master and slave link positions, the velocities and accelerations are estimated by mean of model-based nonlinear observers.

It has been proved that the proposed control law yields local uniformly ultimately boundedess of the closed-loop errors. It is also shown that the final bound of the errors depends on the fourth derivative of the master robot trajectories. A tuning gain procedure to guarantee the stability result has been summarized.

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#### Appendix 1

**Proof of Proposition 1.** Consider a state space representation of (1)–(3), with states  $q_s$ ,  $\dot{q}_s$ ,  $\theta_s$ ,  $\dot{\theta}_s$ , then from the observer (23), (24) and the joint estimation errors (25), it follows that:

$$\tilde{q} = \dot{q} - \mu_1 \tilde{q}$$

$$\tilde{\tilde{q}} = -M_s^{-1}(q_s)((2C_s(q_s, \dot{q}_s) - C_s(q_s, \tilde{q}))\tilde{q} + K_s(\tilde{q} - \tilde{\theta})) - \mu_2 \tilde{q}$$

$$\tilde{\tilde{\theta}} = \tilde{\theta} - \mu_3 \tilde{q}$$
(56)

$$\dot{\tilde{\theta}} = -J_s^{-1}K_s(\tilde{\theta}-\tilde{q}) - J_s^{-1}B_{v,s}\tilde{\theta} - \mu_4\tilde{q}$$

The first and third equation of (56) imply that:

$$\widetilde{\dot{q}} = \widetilde{\dot{q}} + \mu_1 \widetilde{q}, \quad \widetilde{\ddot{q}} = \widetilde{\dot{q}} - \mu_1 \widetilde{\dot{q}}$$
(57)

$$\tilde{\vec{\theta}} = \tilde{\vec{\theta}} + \mu_3 \tilde{q}, \quad \tilde{\vec{\theta}} = \tilde{\vec{\theta}} - \mu_3 \tilde{\vec{q}}$$
(58)

therefore (16) and (56) yield the joint estimation error dynamics (28).

Consider the joint estimation errors given by (25), and introduce the variables  $\tilde{\vec{q}}$ ,  $\tilde{q}^{(3)}$  as:

$$\widetilde{\widetilde{q}} = \widetilde{q}_s - \widehat{\widetilde{q}}_s, \quad \widetilde{q^{(3)}} = q_s^{(3)} - \widehat{q_s^{(3)}}$$
(59)

Differentiating (1) twice, and by considering (2),  $\tau_s$  given by (17), and property (4) it follows that:

$$M_{s}(q_{s})(q_{s}^{(4)}-v(t))+\Psi(q_{s},\dot{q}_{s},\ddot{q}_{s},q_{s}^{(3)},\tilde{q},\tilde{\tilde{q}},\tilde{\tilde{q}},\tilde{\tilde{q}},\tilde{\tilde{q}},\tilde{\tilde{\theta}},\tilde{\tilde{\theta}},\tilde{\tilde{\theta}})=0$$
(60)



Fig. 3 Estimation link and motor position error  $\tilde{q}$ ,  $\tilde{\theta}$ 

where  $\Psi(q_s, \dot{q}_s, \ddot{q}_s, q_s^{(3)}, \tilde{q}, \tilde{q}, \tilde{q}, \tilde{q}, \tilde{q}, \tilde{\theta}, \tilde{\theta})$  represents the mismatch with the desired linearization, which is caused by absence of the high order derivatives of  $q_s$ ,  $q_m$ ,  $\theta_s$ .

According to the definition of  $\ddot{q}$  (59), and by considering (11), (18) and (57) it follows that:

$$\begin{split} \widetilde{\widetilde{q}} &= -M_s^{-1}(q_s)((2C_s(q_s,\dot{q}_s) - C_s(q_s,\dot{\widetilde{q}} + \mu_1\widetilde{q}))(\dot{\widetilde{q}} + \mu_1\widetilde{q}) \\ &+ K_s(\widetilde{q} - \widetilde{\theta})) \end{split} \tag{61}$$

In a similar way, but considering (59), (12) and (19), it is obtained that:

$$q^{(\overline{3})} = -M_s^{-1}(q_s)(\dot{M}_s(q_s,\dot{q}_s)\ddot{q}_s - \dot{M}_s(q_s,\hat{q}_s)\hat{q}_s + \dot{N}(q_s,\dot{q}_s,\dot{q}_s)$$
$$-\dot{N}(q_s,\hat{q}_s,\hat{q}_s) + K_s(\tilde{q} - \tilde{\theta}))$$
(62)

where, by considering property (4) and after a straightforward computation,

 $\dot{N}(q_{s},\dot{q}_{s},\ddot{q}_{s}) - \dot{N}(q_{s},\hat{q}_{s},\hat{q}_{s})$ 

$$\dot{M}_{s}(q_{s},\dot{q}_{s})\ddot{q}_{s}-\dot{M}_{s}(q_{s},\hat{q}_{s})\hat{q}_{s}=\frac{\partial M_{s}(q_{s})}{\partial q_{s}}(\tilde{q}\ddot{q}_{s}+(\dot{q}_{s}-\tilde{q})\tilde{q})$$
(63)

$$= C_{s}(q_{s},\tilde{q})\ddot{q}_{s} + (C_{s}(q_{s},\dot{q}_{s}) - C_{s}(q_{s},\tilde{q}))\ddot{q} + \frac{\partial g_{s}(q_{s})}{\partial q_{s}}\ddot{q}$$

$$+ \begin{bmatrix} C_{s1}(q_{s}) \\ \vdots \\ C_{sn}(q_{s}) \end{bmatrix} (\ddot{q}\dot{q}_{s} + (\ddot{q}_{s} - \ddot{q})\tilde{q})$$

$$+ \dot{q}_{s}\frac{\partial}{\partial q_{s}} \begin{bmatrix} C_{s1}(q_{s}) \\ \vdots \\ C_{sn}(q_{s}) \end{bmatrix} (\ddot{q}\dot{q}_{s} + (\dot{q}_{s} - \ddot{q})\tilde{q})$$

$$+ \ddot{q}\frac{\partial}{\partial q_{s}} \begin{bmatrix} C_{s1}(q_{s}) \\ \vdots \\ C_{sn}(q_{s}) \end{bmatrix} ((\dot{q}_{s} - \ddot{q})(\dot{q}_{s} - \ddot{q})) \qquad (64)$$

Let  $\Phi(q_s, \dot{q}_m, \ddot{q}_m, q_m^{(3)}, \dot{e}, \ddot{e}, e^{(3)}, \tilde{q}, \ddot{\tilde{q}}, \tilde{\theta}, \dot{\tilde{\theta}})$  denotes the function  $\Psi(q_s, \dot{q}_s, \ddot{q}_s, q_s^{(3)}, \tilde{q}, \tilde{\tilde{q}}, \tilde{q}, \tilde{q}, \tilde{q}, \tilde{\theta}, \tilde{\theta})$  after substitution of the relations (57), (58), (61), (62), (63), (64), and (16), it holds that

$$\Phi(q_{s},\dot{q}_{m},\ddot{q}_{m},q_{m}^{(3)},\dot{e},\ddot{e},e^{(3)},\widetilde{q},\widetilde{q},\widetilde{\theta},\widetilde{\theta})$$
$$=\Psi(q_{s},\dot{q}_{s},\ddot{q}_{s},q_{s}^{(3)},\widetilde{q},\widetilde{q},\widetilde{q},\widetilde{q},\widetilde{q}^{(3)},\widetilde{\theta},\widetilde{\theta})$$
(65)

where  $\Phi(q_s, \dot{q}_m, \ddot{q}_m, q_m^{(3)}, \dot{e}, \ddot{e}, e^{(3)}, \tilde{q}, \tilde{q}, \tilde{\theta}, \tilde{\theta})$  is the result of a straightforward chain of substitutions and simplifications. Substitution of (65) and (20) in (60), and considering (21), yields the synchronization error dynamics (26).

Consider the observer (21) and the estimation synchronization errors (22), then it follows that:

$$\hat{e}^{(4)} - \Gamma_1 \tilde{e}^{(3)} - \Gamma_2 \ddot{\tilde{e}} - \Gamma_3 \dot{\tilde{e}} - \Gamma_4 \tilde{e} = 0$$
(66)

Subtraction of (66) from the synchronization error dynamics (26), and considering the estimation synchronization errors (22), yields the estimation synchronization error dynamics (27).

# Appendix 2

**Proof of Lemma 6.** From the properties of the matrices  $M_s(q_s)$ ,  $C_s(q_s, \dot{q}_s)\dot{q}_s$ , the gravity term  $g_s(q_s)$ , (see Section 2), and because their nonlinear terms contain only sinusoidal functions of  $q_s$ , we have that for all  $q_s \in \mathbb{R}^n$ , their partial derivatives can be bounded as:

$$\left\| \frac{\partial M_{s}(q_{s})}{\partial q_{s}} \right\| \leq M_{pM}, \quad \left\| \frac{\partial^{2} M_{s}(q_{s})}{\partial q_{s}^{2}} \right\| \leq M_{ppM}$$

$$\left\| \frac{\partial g_{s}(q_{s})}{\partial q_{s}} \right\| \leq G_{pM}, \quad \left\| \frac{\partial^{2} g_{s}(q_{s})}{\partial q_{s}^{2}} \right\| \leq G_{ppM}$$

$$\left\| \begin{bmatrix} C_{s1}(q_{s}) \\ \vdots \\ C_{sn}(q_{s}) \end{bmatrix} \right\| \leq C_{qM},$$

$$\frac{\partial}{\partial q_{s}} \begin{bmatrix} C_{s1}(q_{s}) \\ \vdots \\ C_{sn}(q_{s}) \end{bmatrix} \right\| \leq C_{pM}, \quad \left\| \frac{\partial^{2} g_{s}^{2}}{\partial q_{s}^{2}} \begin{bmatrix} C_{s1}(q_{s}) \\ \vdots \\ C_{sn}(q_{s}) \end{bmatrix} \right\| \leq C_{ppM}$$

For the sake of simplicity and without loss of generality, let assume that  $Q_x = -(P_x A + A^T P_x)$  is a symmetric positive definite block diagonal inatrix, with  $n \times n$  block entries, and denote the *i*-th diagonal  $n \times n$  block of  $Q_x$  by  $Q_{xi}$ . Then from the definition of  $x_N$ ,  $y_N$ , and  $\xi_N$  (47), and V given by (43), it follows that the term  $x^T Q_x x$  in (43) can be bounded as  $x_N^T R_x x_N$ , with  $R_x \in \mathbb{R}^{8 \times 8}$ 

$$R_x = \text{diag}\{Q_{xiM}\}$$
  $i = 1, \dots, 8$  (67)

where  $Q_{xiM}$  is the maximum eigenvalue of  $Q_{xi}$ , and such that positive definiteness of  $Q_x$  implies that  $R_x$  is positive definite.

From  $Q_y$  (44), the term  $\Omega$  (46), and the bounds of the partial derivatives of  $M_s(q_s)$ ,  $C_s(q_s, \dot{q}_s)\dot{q}_s$ , and  $g_s(q_s)$ , it follows that  $y^T Q_y y$  can be bounded as  $y_N^T R_y y_N$ , with  $R_y \in \mathbb{R}^{4 \times 4}$ 

$$R_{y} = \begin{bmatrix} \alpha_{0}^{*} & \alpha_{1}^{*} & \alpha_{2}^{*} & \alpha_{3}^{*} \\ \alpha_{1}^{*} & M_{sm}\mu_{1m} - 2\lambda_{0} + 2C_{M}V_{M} & \eta_{0}\mu_{3M} - \frac{1}{2}K_{sM} & \frac{1}{2}J_{sM}\mu_{3M} \\ \alpha_{2}^{*} & \eta_{0}\mu_{3M} - \frac{1}{2}K_{sM} & 2\eta_{0}J_{sM}^{-1}K_{sm} & \alpha_{4}^{*} \\ \alpha_{3}^{*} & \frac{1}{2}J_{sM}\mu_{3M} & \alpha_{4}^{*} & B_{v,sm} - 4\eta_{0} \end{bmatrix}$$

$$\alpha_{0}^{*} = 2\lambda_{0}(M_{sM}^{-1}(K_{sm} + 2C_{M}V_{M}\mu_{1m}) + \mu_{2m}), \quad \alpha_{1}^{*} = C_{M}V_{M}(2M_{sm}^{-1}\lambda_{0} + \mu_{1M}) - \frac{1}{2}M_{sM}\mu_{2M} - \frac{1}{2}\beta_{1} \\ \alpha_{2}^{*} = -\lambda_{0}M_{sm}^{-1}K_{sM} + \eta_{0}(\mu_{4M} + J_{sm}^{-1}B_{v,sM}\mu_{3M} - J_{sm}^{-1}K_{sM}), \\ \alpha_{3}^{*} = \frac{1}{2}(B_{v,sM}\mu_{3M} + J_{sM}\mu_{4M} - K_{sM}), \quad \alpha_{4}^{*} = \eta_{0}J_{sm}^{-1}B_{v,sM} - \frac{1}{2}\mu_{4M} - \frac{1}{2}\beta_{2} \end{bmatrix}$$

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(68)

The matrices  $R_x$  and  $R_y$  are related to the bounds of the quadratic terms in x and y of V (43). Nevertheless, there exist other cross quadratic terms in x and y which come from  $Q_{xy}$  (45) and  $\Omega$ . These cross quadratic terms are bounded, such that,  $R_{xy}$  corresponds to the bound of the cross quadratic terms of:

$$\begin{aligned} \|x^{T}Q_{xy}x\| + \|x^{T}P_{x}B[-M_{s}^{-1}(\Phi+K_{s}J_{s}^{-1}K_{s}(y_{1}-y_{3}) \\ -K_{s}J_{s}^{-1}B_{v,s}(y_{4}+\mu_{3}y_{1})) + q_{m}^{(4)}]\| \end{aligned}$$

Let consider  $\Phi + K_s J_s^{-1} K_s (y_1 - y_3) - K_s J_s^{-1} B_{v,s} (y_4 + \mu_3 y_1)$ , this term can be bounded as

$$\|\Phi + K_s J_s^{-1} K_s (y_1 - y_3) - K_s J_s^{-1} B_{v,s} (y_4 + \mu_3 y_1)\| \leq \Phi_1 + \Phi_r$$
(69)

where  $\Phi_1$  contains terms of first order in x, y, and  $\Phi_r$  contains the remaining terms (orders 2, 3 and 4). After a long straightforward computation it is obtained that  $\Phi_1$  is given by:

$$\Phi_1 = ((a_1 + a_3)\mu_{1M} + a_2 + a_3\mu_{3M}) \|y_1\| + (a_1 + a_3) \|y_2\| + a_2 \|y_3\| + a_3 \|y_4\| a_1 = 2M_{-1}^{-1} C_M V_M [A_M(M_{DM} + 2C_{DM}) + G_{DM} + K_{DM} + V_{2M}^2 (M_{DM})]$$

$$\begin{aligned} &+2C_{pM})]+C_{pM}V_{M}(6A_{M}+2M_{sm}^{-1}C_{M}V_{M}^{2})+4C_{ppM}V_{M}^{3}\\ &+2(A_{M}^{2}+C_{pM}A_{M}V_{M}+2M_{sm}^{-1}C_{M}V_{M}^{2})+4C_{ppM}V_{M}^{3}\\ &+2(A_{M}^{2}+C_{pM}A_{M}V_{M}+2M_{sm}^{-1}C_{pM}C_{M}V_{M}^{2})\\ &+M_{sm}^{-1}V_{M}(2M_{pM}+C_{M}+C_{qM})[G_{pM}+3C_{pM}V_{M}^{2}+(M_{pM}+C_{M}+C_{qM})]\\ &+C_{M}+C_{qM})(A_{M}+2M_{sm}^{-1}V_{M}^{2})]+D_{M}(2M_{pM}+C_{M}+C_{qM})\\ &+2M_{sm}^{-1}C_{M}V_{M}A_{M}(M_{pM}+2C_{qM})+2G_{ppM}+2M_{ppM}A_{M}V_{M}\\ &a_{2}=K_{sm}M_{sm}^{-1}\{2A_{M}(M_{pM}+2C_{qM})+G_{pM}+K_{sM}+V_{M}^{2}(M_{pM}+2C_{pM})\}\\ &+(M_{sm}^{-1}V_{M})^{2}(M_{pM}+C_{M}+C_{qM})(2M_{pM}+C_{M}+C_{qM})\}\\ &a_{3}=K_{sm}M_{sm}^{-1}V_{M}(2M_{pM}+C_{M}+C_{qM})\end{aligned}$$

Therefore, the matrix  $R_{xy}$  is given by:

$$R_{xy} = M_{sm}^{-1} P_{Q} [R_{1}^{*} \ a_{1} + a_{3} \ a_{2} + K_{sM}^{2} J_{sm}^{-1} \ a_{3} + K_{sM} J_{sm}^{-1} B_{v,sM}]$$

$$P_{Q} = \begin{bmatrix} (P_{x14} + P_{x18})_{M} \\ (P_{x24} + P_{x28})_{M} \\ (P_{x34} + P_{x38})_{M} \\ (P_{x44} + P_{x48})_{M} \\ (P_{x45} + P_{x58})_{M} \\ (P_{x46} + P_{x68})_{M} \\ (P_{x47} + P_{x78})_{M} \\ (P_{x48} + P_{x88})_{M} \end{bmatrix}$$

$$(70)$$

$$R_1^* = (a_1 + a_3)\mu_{1M} + a_2 + a_3\mu_{3M} + K_{sM}J_{sm}^{-1}(B_{v,sM}\mu_{3M} - K_{sm})$$

**Remark 12**. Notice that  $a_1$ ,  $a_2$ ,  $a_3$  are uniquely determined by the physical parameters of the slave robot and the bounds of the master trajectories, and thus they do not depend on the control and observer gains. As a consequence  $a_1$ ,  $a_2$ ,  $a_3$  must be evaluated only once.

At this point all the quadratic terms of V have been bounded in terms of  $R_x$ ,  $R_y$ ,  $R_{xy}$ , and  $x_N$ ,  $y_N$ ,  $\xi_N$ . Therefore it is only necessary to bound all the remaining terms originated from  $\Omega$ . From (46) and considering (69), it follows that:

$$\|\Omega(q_{s},\dot{q}_{m},\ddot{q}_{m},q_{m}^{(3)},q_{m}^{(4)},x,y)\| \leq \Theta(V_{M},A_{M},D_{M},E_{M},\xi_{N})$$
(71)

where by considering  $P_Q$ , (70),  $\Theta(V_M, A_M, D_M, E_M, \xi_N)$  is given by:

$$\begin{split} \Theta &= 2x_N^T P_Q [M_{sm}^{-1} \Phi r + E_M] + 2\lambda_0 C_M M_{sm}^{-1} \|y_1\| \|y_2\| (\|y_2\| \\ &+ \mu_{1M} \|y_1\|) + C_M \mu_{1M} \|y_1\| \|y_2\|^2 + C_M (\mu_{1M} \|y_1\| + 2\|x_2\|) \\ &\times (2\lambda_0 M_{sm}^{-1} \|y_1\| + \|y_2\|) (\|y_2\| + \mu_{1M} \|y_1\|) \end{split}$$

## Appendix 3

**Proof of Proposition 9.** First notice that the definition of  $\beta_1$ ,  $\beta_2$  given by (54), imply that  $\alpha_1^* = 0$ ,  $\alpha_4^* = 0$  in  $R_y$  (68). Second, let  $R_{vi}$  denote the determinant of the *i*-th leading minor of  $R_v$ , then conditions for  $R_{vi} > 0$ , i = 1, ..., 4, are given by:

- *R<sub>y1</sub>*>0 if λ<sub>0</sub>>0, μ<sub>1m</sub>>0, and μ<sub>2m</sub>>0 *R<sub>y2</sub>*>0 if μ<sub>1m</sub>>2*M*<sup>-1</sup><sub>sm</sub>(λ<sub>0</sub>-*C<sub>M</sub>V<sub>M</sub>)
  For <i>R<sub>y3</sub>*, first notice that it can be written as *R<sub>y3</sub>=a<sub>31</sub>μ<sub>2m</sub>*  $+a_{30}$ , with  $a_{31}=b_{32}\mu_{3M}^2+b_{31}\mu_{3M}+b_{30}$ , and  $b_{32}<0$ . Then  $a_{31} > 0$  if  $\mu_{33} < \mu_{3m}$ ,  $\mu_{3M} < \mu_{33}$ , where

$$\mu_{33_{-}}, \mu_{33}^{-} = \frac{1}{4J_{sM}\eta_0} (2J_{sM}K_{sM} \mp 4(2J_{sM}\eta_0K_{sM}(M_{sm}\mu_{1m} + 2(C_M V_M - \lambda_0)))^{1/2})$$

because  $a_{31} > 0$ , then  $\mu_{2m} > -a_{31}^{-1}a_{30}$  implies  $R_{y3} > 0$ . •  $R_{y4}$  can be written as  $R_{y4} = a_{41}\mu_{2m} + a_{40}$ , with  $a_{41}$  $=b_{42}\mu_{3M}^2+b_{41}\mu_{3M}+b_{40}$ , and  $b_{42}<0$  if  $\eta_0$  holds:

$$0 < \eta_0 < \min\left\{\frac{B_{v,sm}}{4}, \frac{1}{8J_{sM}}(J_{sM}B_{v,sm} + (J_{sM}^2B_{v,sm}^2 + 8J_{sM}J_{sm}^2K_{sm})^{1/2})\right\}$$

Then  $b_{42} < 0$  implies that  $a_{41} > 0$  if  $\mu_{34} < \mu_{3m}$ ,  $\mu_{3M} < \mu_{34}$ , where

$$\mu_{34_{-}}, \ \mu_{34}^{-} = \frac{1}{2(\eta_0(2J_{sM}(4\eta_0 - B_{v,sm})) - J_{sm}^2 K_{sm})} \times (2J_{sM}K_{sm}\eta_0(4\eta_0 - B_{v,sm}) \pm (2J_{sM}K_{sm}\eta_0)^{1/2} \cdot ((4\eta_0 - B_{v,sm})[J_{sm}^2 K_{sM}^2 + 8\eta_0(2\lambda_0 - 2C_M V_M - M_{sm}\mu_{1m})(J_{sM}^{-1}J_{sm}^2 K_{sm} - 2\eta_0(4\eta_0 - B_{v,sm}))])^{1/2})$$

because  $a_{41} > 0$ , then  $\mu_{2m} > -a_{41}^{-1}a_{40}$  implies  $R_{v4} > 0$ .

If the above conditions are satisfied, then the determinants of all the leading minors of  $R_y$  are positive. Therefore from the Sylvester's criterion, it follows that  $R_{y}$  is positive definite.

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